Kafr elsheikh University
Faculty of Engineering
Mechanical Power Eng. Department
4<sup>th</sup> Year Power Mechanics
Subject: Turbo-machinery
Examiner: Dr. M.I. Amro and commit



Date: 3/1/2019
Time Allowed: 3 Hours
Full Marks: 75
Final Exam
No. of Pages: 3 page
Academic Code: MEP427

# يسمتح باستخدام جداول و خرانط البخار

Assume any missing data, use schematic diagrams in your answer

#### Question (1) (25 marks)

1-a) Assigned if the following statements is correct or incorrect and correct the false

(15 marks)

- 1- In a reaction turbine, the relative velocity coefficient is less than unity
- 2- Blade loading coefficient is used to express work capacity of the stage.
- 3- Radial flow turbine is used in all applications of the gas turbine power plant
- 4- The axial flow turbine has been used exclusively in aircraft gas turbine engines
- 5- In an impulse turbine, pressure drop of expansion takes place in the stator as well as rotor
- 6- It is much more difficult to arrange for an efficient deceleration of flow than it is obtain an efficient acceleration.
- 7- Speed of turbines must be governing and always the gears box is fixed after its stages.
- 8- In multi-stages turbine, the same blades shape can be used if the absolute velocity at inlet and outlet of stage is the same; in addition the inlet and outlet flow angle is equally.
- 9- In a Rateau turbine, every turbine stage has a row of fixed Convergent-Divergent nozzle and the pressure energy is divided between all turbine stages to reduce the turbine velocity.
- 10-In impulse turbines, the moving blades convert the pressure energy into rotational energy and direct the gases flow to the next stages.
- 11- The turbine is more efficient than the compressor since the expansion of gases save the boundary layer and does not permits, to the flow separation occurred.
- 12- In a reaction turbine, the blades passage area varies continuously to allow for the continued expansion of the gas stream over the rotor blades.
- 13- In case of maximum blade efficiency in a single stage impulse turbine, the steam leaves the blades axially with no axial thrust.
- 14- In multi-stage steam turbines, the impulse turbine stage is used in last stage and the reaction stages are used where the pressure drop per stage is high.
- 15- Curtis turbine is a pressure-compound impulse turbine, while Rateau turbine is a velocity-compound impulse turbine.

1-b) At a stage in a reaction turbine, the mean blade ring diameter is 1 m and the turbine runs at a speed of 50 rev/s. The blades are designed for 50% degree of reaction with exit angles of 30° and inlet angles of 50°. The turbine is supplied with steam at the rate of 600 tons/hr. and the stage efficiency is 85%. Determine:

- a) The power output of the stage, and the specific enthalpy drop in the stage,
- b) The percentage increase in relative velocity in the moving blades due to expansion in these blades.

#### Question (2) (20 marks)

- 2-a) Draw and describe Rateau and Curtis turbines, indicate the velocity and pressure distributions through each turbine. (5 marks)
- 2-b) Prove that; for the equiangular frictionless blades of impulse turbine; the blade or diagram maximum efficiency is equal to:  $\eta_{b,max} = \cos^2 \alpha_1$ ,

Draw the velocity compound diagram at that condition

(5 marks)

- 2-c) A single stage impulse turbine is to be designed that the initial steam conditions are 7 bar and 260° C and the exhaust pressure is 1 bar. The nozzles angle is to be 15° to the plane of rotation. Nozzles efficiency is 95 %. Blade exit angle is 30° and the blade velocity coefficient is 85%. The mean diameter of wheel is 1100mm.
  - a) What is the blade speed ratio when the wheel speed is 6000 rpm?
  - b) How much power is delivered to the wheel at this speed of 6000 rpm when the steam flow is 2250 kg/hr.?
  - c) At what wheel speed would the maximum power be developed?
  - d) What would be the maximum power?

(10 marks)

### Question (3) (10marks)

Steam is supplied to a condensing turbine at 17 bar and 330 C°. Condenser pressure is 0.095 bar. Steam is tapped off at one point for extraction to a certain process, where the extraction pressure equals 1.6 bar. The turbine delivers 2000 kW at full load, with an internal efficiency of 80 %. When there is a steam process available at full load, all the steam is extracted at 1.6 bar.

If the no load steam consumption in both no extraction and extraction cases equals 12 % of the full load consumption, and the mechanical and generation efficiencies are 96.2% and 99 % respectively.

- a) Calculate the steam consumption at no load and full load for both cases of no extraction and extraction.
- b) Draw Willans line for each case.
- c) How much steam will be generated in the boiler for a power output of 1500 kW, if 6500 kg/hr. of steam are extracted from the turbine? (10 marks)

# Question (4) (20marks)

S Yanamera se

- 4-a) Derive a relation for calculating the degree of reaction and the blade loading coefficient as a function on inlet and outlet blade angles ( $\beta_2$  and  $\beta_3$ ) for axial flow gas turbine? For a degree of reaction A=0.5 prove that:  $\psi=4\phi\tan\beta_3-2$  and  $\psi=4\phi\tan\beta_2+2$  (5marks)
- 4-b) Draw and describes the velocity triangles and (T-S) diagram for the axial flow turbine indicating the values of pressure and velocity at all points.
- 4-c) In a single-stage axial flow gas turbine gas enters at stagnation temperature of 1100K and stagnation pressure of 5 bars. Axial velocity is constant through the stage and equal to 250 m/s. Mean blade speed is 350 m/s. Mass flow rate of gas is 15 kg/s and assume equal inlet and outlet velocities. Nozzle efflux angle is 63°, stage exit swirl angle equal to 9°. Determine the rotor-blade gas angles, degree of reaction, and power output. (10 Marks)

With Best Wishes

Dr. Mohamed Amro and commit